

## Effect of fatigue crack orientation on the sensitivity of eddy current inspection in martensitic stainless steels

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## Outline



#### Introduction

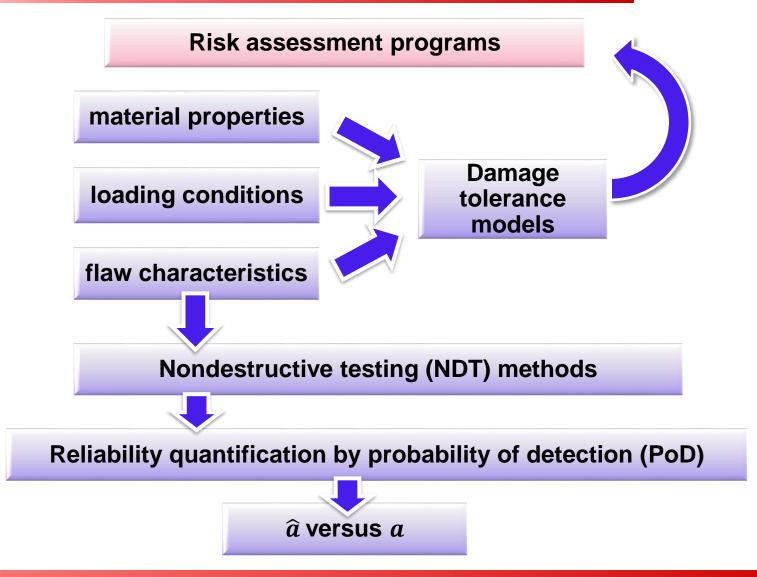
Experimental procedure and data analysis

**FE modelling and analysis** 

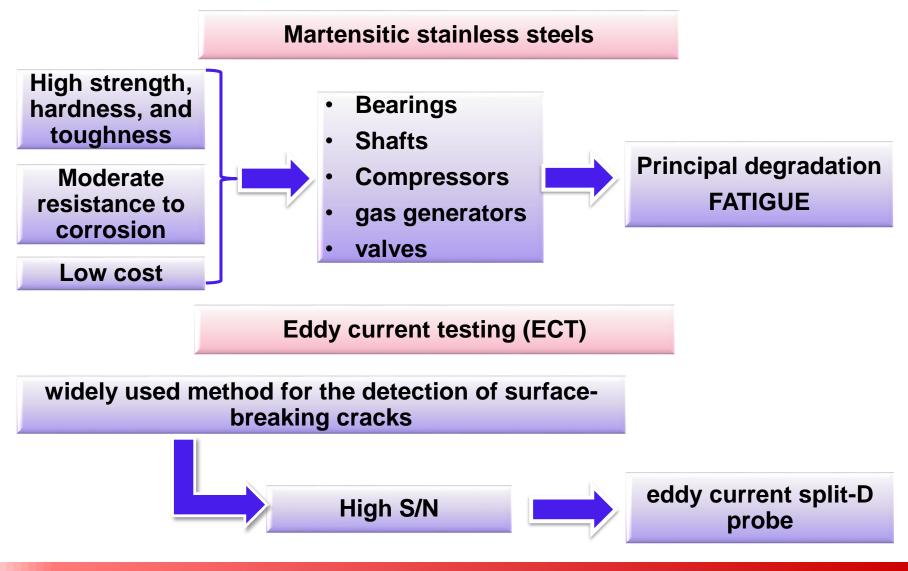
Results and discussion

### Conclusions



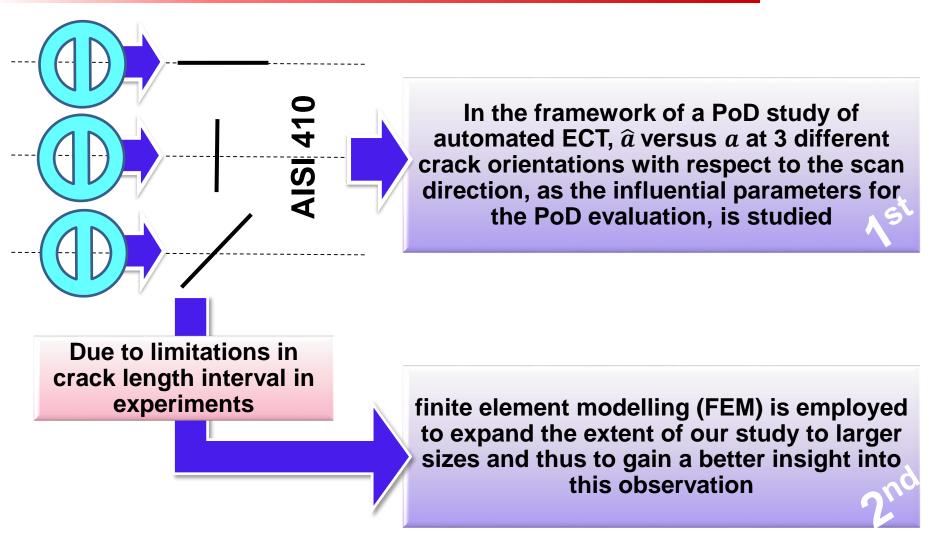






## Objectives









**Experimental procedure and data analysis** 

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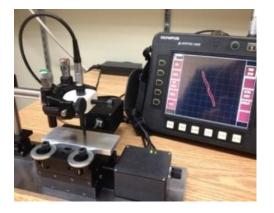
Results and discussion

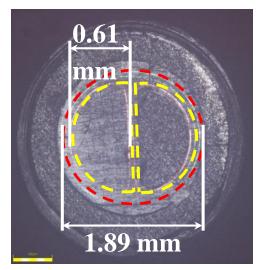
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## Test unit and samples used in the study

ÉTS

- Nortec 500S along with a reflection differential split-D probe are used
- The probe's frequency range is 500kHz-3Mhz
- Starter flaws using electrical discharge machining (EDM) process on the surface of samples
- cyclically loaded in order to grow fatigue cracks out of the starter flaws.
- Samples containing fatigue cracks of 0.76 to 2.95 mm in length
- According to destructive tests, Their depth linearly increases with their length.



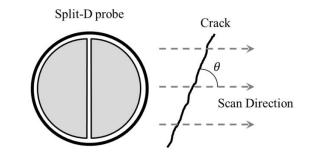


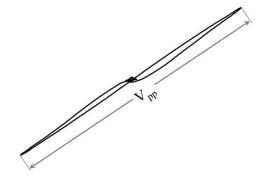
#### Depth = -0.0006 + 0.3475 Length, $r^2 = 0.9845$

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## ECT automated scans and signal analysis

- Calibration on a reference flaw:
  - device gain
  - the impedance plane angle
  - perpendicularity of the probe to the sample's surface
- Initial lift-off of 0.03 mm
- Raster scans at frequencies of 500 kHz and 1 MHz
- Scan index of 0.5 mm
- Gains are compensated for each axis
- signals' characteristics are extracted
- **ECT** signal length  $(V_{pp})$ : main parameter under study











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## 3D model and material properties used for the assembly of the probe and sample

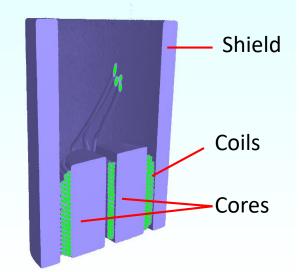


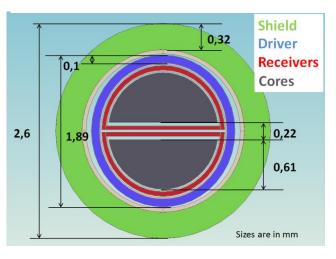
#### 3-D modeling in Comsol multyphysics:

- A half-scaled CAD model for orientations of 0° and 90° owing to their plane symmetry
- Full model for orientation of 45°
- Cracks represented by semi-elliptical notches having 0.02 mm opening
- Dimensions of the probe's Interior components according to X-ray tomography reconstruction
- Initial lift-off of 30 μm

#### Material properties: measurements and data sheets

Component	Relative permeability	Electrical conductivity
Cores and shield	2500	1(S/m)
Sample	300	1.9e6(S/m)





## Physics, mesh and solver



#### Physics:

- MF physics within AC/DC module
- Multi turn domains for coils
- Magnetic insulation boundary condition for encompassing air domain

$$(\nabla \times (\nabla \times \mathbf{A})) / \mu_0 \mu_r + (j\sigma\omega - \omega^2 \varepsilon_0 \varepsilon_r)\mathbf{A} = \mathbf{J}_e \quad j = \sqrt{-1}$$

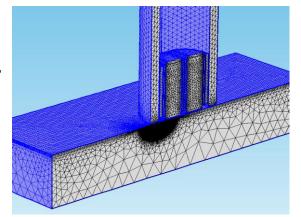
Mesh:

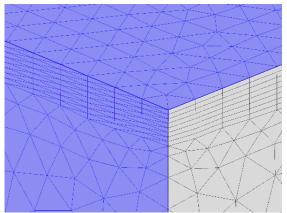
 $\Delta \mathbf{Z} = (\mathbf{V}_{R2} - \mathbf{V}_{R1}) / \mathbf{I}_D$ 

- Second order tetrahedral elements
- 8 boundary layer mesh on the surface of the sample
- Each layer has the thickness of first standard penetration depth
- Finer elements for the notch geometry
- Solver:



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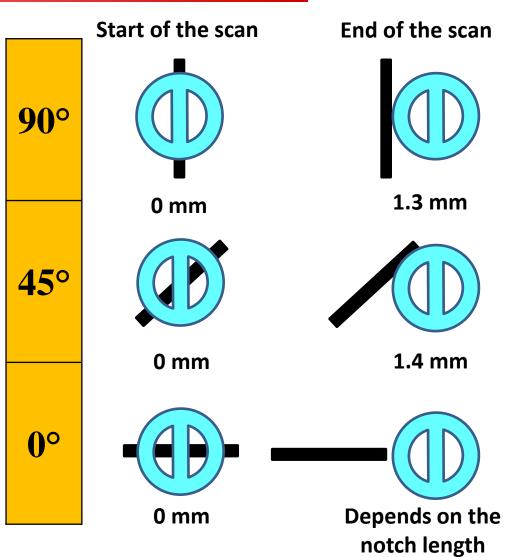
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## Details of simulated scans



- 1.3 mm displacements of the probe along the scan path
- Probe's displacement increments of 0.1 mm
- Probe is centered by the notch at the beginning of the scan

Orientation	Notch length Variations(mm)	Steps (mm)
<b>90°</b>	2 - 6	0.5
45°	1.5 - 4.5	0.5
<b>0</b> °	0.5 - 3	0.5







Experimental procedure and data analysis

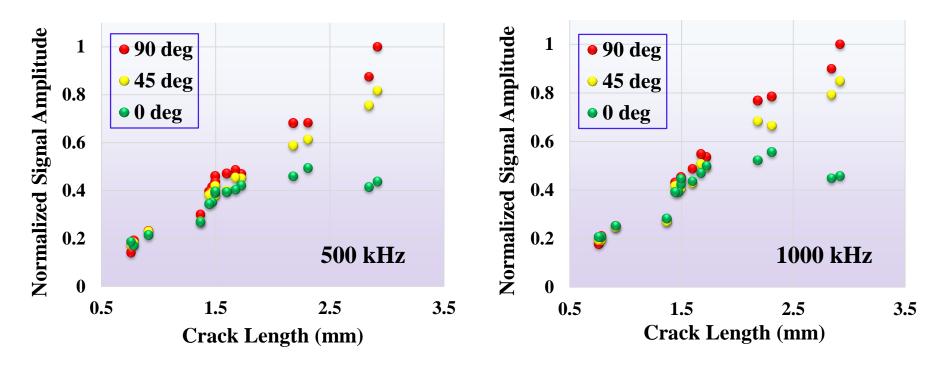
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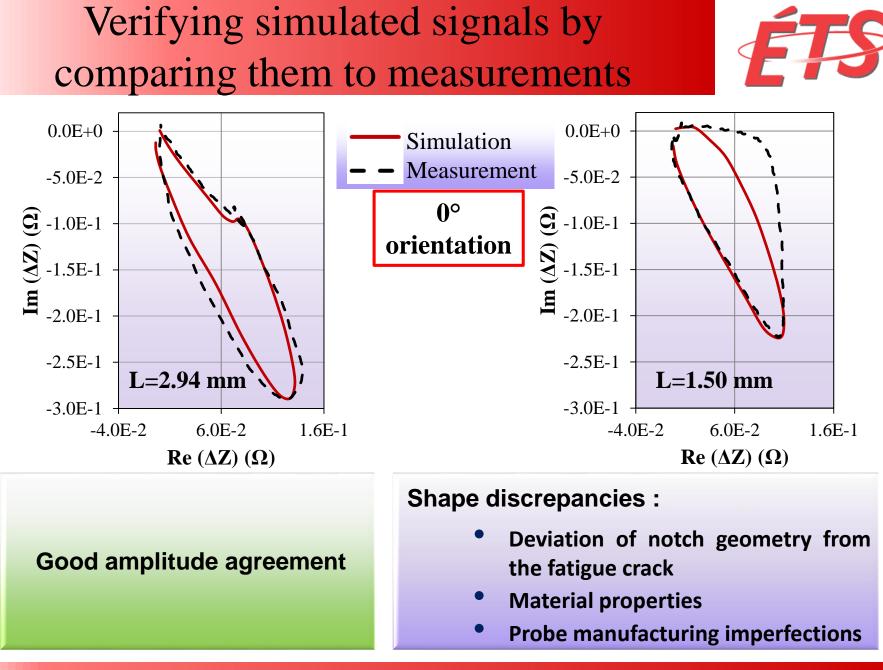
#### Conclusions

# Measured signal amplitude versus crack length





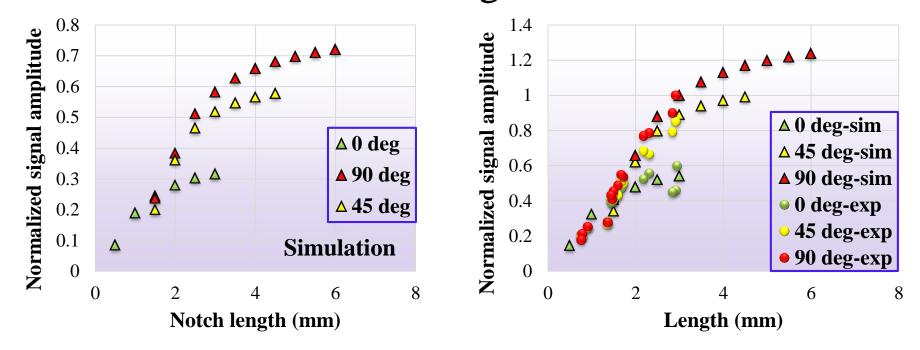
- ❑ The same behaviour at both frequencies
- □ Amplitude is independent of the orientation for crack length below 1.8 mm
- Amplitude changes versus length variation becomes plateau after 1.8 mm for 0° orientation



## Simulated and measured signal amplitude



versus crack length

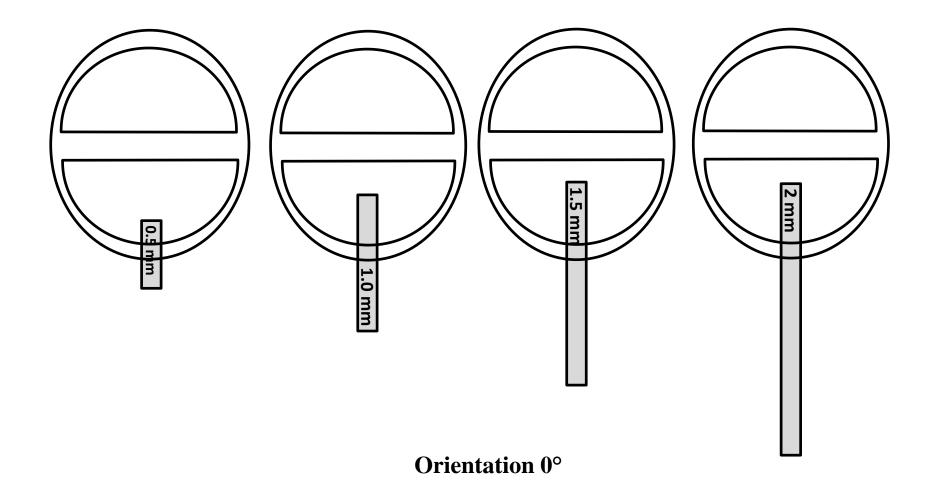


For each orientation, Amplitude variations becomes less than 5 % after a certain notch length:

- 0°- 2.5 mm
- 45°- 3.5 mm
- 90°- 4 mm

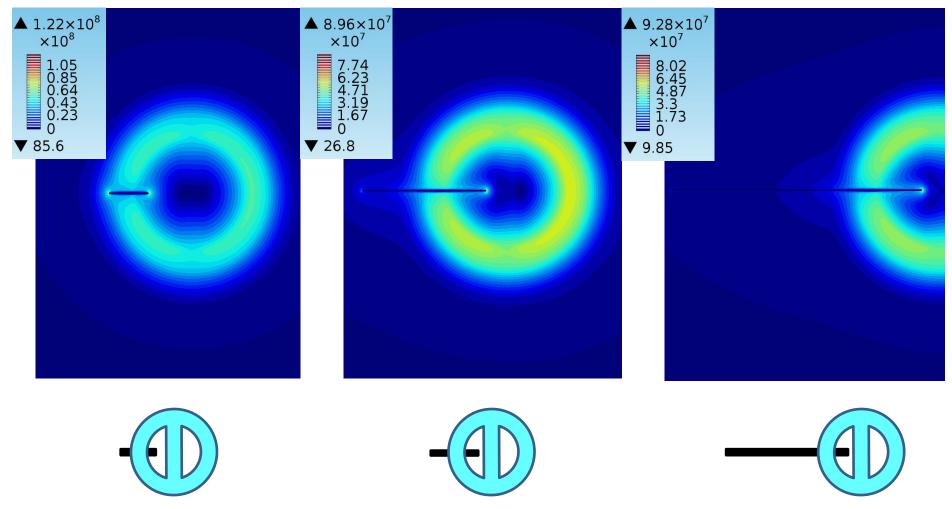
Relative position of the probe and notches at which the amplitude is maximum





## Induced current density distribution as the notch length varies - 0°





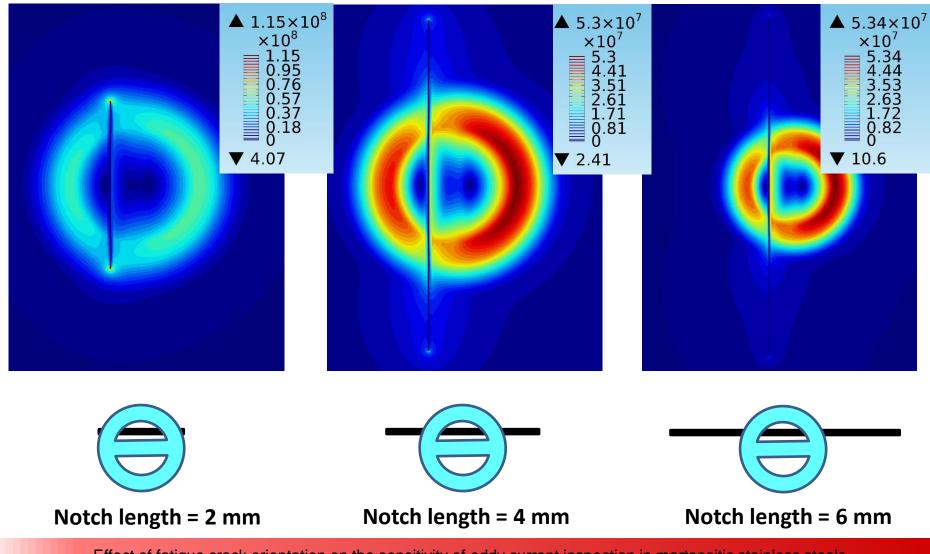
Notch length = 0.5 mm

Notch length = 1.5 mm

Notch length = 3 mm

Induced current density distribution as the notch length varies - 90°









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- Depending on the flaw orientation, the signal amplitude increases with the crack length up to a critical (flaw length)/(drive coil diameter) ratio (L/D) which is specific to the flaw orientation.
- This critical ratio grows as the orientation increases from 0° to 90°
- The variation of the signal amplitude as the crack length increases is almost independent of the crack orientation until a L/D value equal to the unity and then the slope of amplitude versus crack length variations becomes orientation dependent.
- Accordingly, the probability of detection of fatigue is almost independent of crack orientation once the L/D ratio is below the unity.
- The results of our FEM study show a good agreement with the measurements outcome in terms of amplitude. Therefore, this model is a reliable means to carry out model-based studies of probability of detection.



# Thank you for your attention